

THERMAL FATIGUE ANALYSIS OF INDUCTION MELTING FURNACE WALL FOR ALUMINA RAMMING MASS

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Abstract: Furnaces are most commonly used for melting of materials. Induction furnaces are more beneficial as no fuel is required. It is a problem to find life of Induction Melting Furnace wall under thermal fatigue. The induction melting furnace wall is made of alumina ramming mass which is one kind of refractory material. The failure occurs due to cyclic thermal stresses. Temperature distribution and thermal stress distribution fields of the induction melting furnace refractory wall were calculated by using ANSYS finite element analysis software based on the physical description of its failure under thermal fatigue conditions. The thermal fatigue life of the refractory wall is required to be found out by means of thermal stresses created inside the refractory wall of induction melting furnace from modified S – log N Curve.

Keywords: Thermal fatigue, induction furnace, finite element analysis, refractory, alumina ramming mass.

1. INTRODUCTION

1.1 FURNACE [1-3]

Furnace is a term used to identify a closed space here heat is applied to a body in order to raise its temperature. The source of heat may be fuel or electricity. Commonly, metals and alloys and sometimes non-metals are heated in furnaces. The purpose of heating defines the temperature of heating and heating rate.

Increase in temperature softens the metals. They become amenable to deformation. This softening occurs with or without a change in the metallic structure. Heating to lower temperatures (below the critical temperature) of the metal softens it by relieving the internal stresses. On the other hand, metals heated to temperatures above the critical temperatures leads to changes in crystal structures and re-crystallization like annealing. Further some metals and alloys are melted, ceramic products vitrified, coals coked, metals like zinc are vaporized and many other processes are performed in Furnaces.

1.2 DESCRIPTION OF PROBLEM

Induction furnaces are widely used in the iron industry for the casting of the different grades of cast iron products. Refractory wall of induction melting furnace is a key component which is used as insulation layer. It is made of ramming mass like alumina, alumina, magnesia etc. The refractory wall is directly influenced by the thermal cycling of the high temperature molten iron in the furnace. Thermal fatigue failure is easy to happen for it because of the larger phase transformation thermal stresses and it has a shorter life. This can cause serious production accidents. Therefore, the service life problem of the refractory wall has always been a focus of attention in the application of this to the industry.

The research on the distribution rule of temperature and thermal stress field and on the fatigue life assessment method for the refractory wall will not only lay foundation for the study on the thermal fatigue of this kind of parts under thermal shock condition of low cycle and high phase transition stresses but also offers effective control for thermal fatigue failure.

Here, ANSYS Software is used for finite element analysis of refractory wall. ANSYS calculates stress distribution across the refractory wall. The aim of this research is to identify points where critical stresses are generated so that we can minimize stresses in those regions and improve life of refractory wall.

2. LITERATURE REVIEW

Xia Zhou, Zhanfei Tanga and Guohui Qu⁴ have done research work on thermal fatigue life of tundish cover. Continuous casting is widely used in the steel industry for the casting of different grades of steel products. Continuous casting tundish is a key intermediate device which is repeatedly used to store molten steel from melting furnace and then various steel section products are made by the continuous casting and rolling process, etc. The tundish cover is directly influenced by the thermal cycling of the high temperature molten steel in the tundish, so thermal fatigue failure is easy to happen for it because of the larger phase transformation thermal stress and it has a shorter life. This is not conducive to thermal insulation, cleanliness and pouring of liquid steel. What's more, this will cause serious production accidents.

Therefore, the service life problem of the tundish cover has always been a focus of attention in the application of this type of components to industry. The research on the distribution rule of temperature and thermal stress field and on the fatigue life assessment method for the tundish cover will not only lay foundation for the study on the thermal fatigue of this kind of parts under thermal shock condition of low cycle and high phase transition stress σ_5 , but also offer effective control for thermal fatigue failure.

To obtain high-quality continuous casting products, many researchers have studied steel flow, heat transfer and slag-steel interaction in tundishes under isothermal and non isothermal conditions by physical and mathematical modelling. Barron-Meza et al.⁶ studied the effects and considerable changes of mixed convection flow in a tundish heated by a plasma torch using water modelling and mathematical simulation approaches. Mathematical simulations of the effects of a varying ladle stream temperature, heat losses and flow control devices on liquid steel flow patterns were reported by Morales et al.⁵ and

Palafox-Ramos et al.⁵ Solhed et al.⁵ proposed a model of a tundish including the molten steel, slag, and refractory phases. The predictions of heat transfer and fluid-flow characteristics were validated by practical measurement data. In addition, fluid flow, heat and mass transfer of liquid steel in a trough type tundish were physically and mathematically simulated by Vargas-Zamora et al.¹¹ using water model. To the knowledge of the authors, neither the tundish nor the tundish cover has so far been included in any modeling efforts. It is obvious though, that thermal fatigue behavior of the tundish system especially the tundish cover will cause many metallurgical issues in the continuous casting process.

Traditional fatigue life prediction is often based on the statistical lifetime calculation of the large numbers of similar products or on lots of physics experiments, while these two methods are both time-consuming and high cost. In recent years, many scholars have also proposed other methods for evaluating fatigue damage such as the crack propagation methods based on the fracture theory^{12–16} and the finite element methods based on the damage mechanics⁵ However, there still exist some problems when the above-mentioned methods applied to fatigue life analysis: firstly a simplified two-dimensional (2D) model is often used in most methods; secondly no kinds of methods can be totally suitable for the life estimation of structures particularly practical structures subjected to accumulated fatigue damage. In practical engineering, the simplified plane model is often unreasonable because of prominent fatigue problems and geometrical configuration of complex structures. So research on an efficient and accurate fatigue life prediction method has great significance to identify the mechanism of materials or structural failure and improve product life.

3. PREPROCESSING OF FEA USING ANSYS

The accurate simulation of Induction melting furnace refractory wall is done for finding out temperature distribution and thermal stress distribution by using proper solving conditions. These solving conditions include initial and boundary conditions, material properties and assumptions etc. Finite Element Analysis using ANSYS was performed to calculate temperature field and stress field caused by application of heat flux caused by heat generation inside the induction melting furnace.

3.1 ANALYSIS MODEL

Based on drawing and dimensions available from Jyoti CNC Limited, a model is developed using Pro E software and radius is given to minimize effects of heat concentration on edges. We had converted this model in iges format which is universal format for all modeling software then it is imported into ANSYS.

3.2 MESH SIZE SELECTION

We have done coupled field dynamic analysis using different mesh size and found that with the increase in mesh density stress is decreasing up to an optimum value but then if we go for more mesh density then stress will increase. We have done three kinds of meshing course, normal and fine meshing. In course meshing, we had selected 2689 nodes and 1197 elements and we had got maximum stress 223 MPa. Then we had selected normal mesh density 10,018 elements and 17,425 nodes and we had got maximum stresses 176 MPa. In very fine meshing, we had selected 79,773 elements and 1,22,459 nodes and we had got 189 MPa maximum stresses. So we understand that we are going away from the answer so that

we had selected normal mesh density. If we go for very fine meshing or very course meshing we will go away from the accuracy. So that we had selected mesh density which will give be more nearer to the actual value. Thus we had found out mesh density.

3.3 INITIAL AND BOUNDARY CONDITIONS, MATERIAL PROPERTIES AND BASIC ASSUMPTIONS

To solve this heat transfer problem of induction melting furnace wall, the following initial and boundary conditions, material properties and basic assumptions are made:

- Refractory Materials for induction furnace wall meets the basic assumptions in the science of mechanics.
- Environmental Temperature is homogeneous at 25° C.
- Ignore the influence of heat convection.
- Ignore the effect of gravity field.
- The surface of induction melting furnace wall is clean.
- The initial temperature of the induction melting furnace is set 25° C and it is agreement with the ambient temperature during solving the problem.
- Heat flux is considered constant for this analysis.
- Scarp material input inside furnace is considered uniform for our analysis.

Calculation of Heat Flux:

- Heat Flux = Heat Generated / Area
- Heat Flux = $Q_g / (2 * \pi * r * l)$
- Heat Flux = $150 \text{kw} / (2 * 3.14 * 0.3715 * 1.21) \text{m}^2$
- Heat Flux = $150 / 2.82 \text{kw} / \text{m}^2$
- Heat Flux = $53,000 \text{watt} / \text{m}^2$

4.0 RESULTS AND DISCUSSION

It gives temperature distribution and stress distribution for four different materials alumina ramming mass. It indicates that temperature at inner side of the furnace wall is higher than at outside of the furnace wall.

We had entered material properties of alumina ramming mass. We had found out temperature distribution after 1 hour, 2 hour and 3 hour. We had plotted a graph of change in temperature with respect to time at inner surface and outer surface of the furnace wall. We had found out stress distribution after 20 minutes, 40 minutes and 60 minutes.

We had plotted a graph of stress variation with respect to time at inner surface and outer surface of furnace wall. The red colour in the stress distribution diagrams indicate maximum stresses created and from that region minor crack propagation will be started for fatigue failure.

Table 1. Table captions should be placed above the table

Material Properties Of Alumina Ramming Mass			Unit
1	Elasticity Constant	222	GPa
2	Poisson's Ratio	0.22	-
3	Density	3400	KG/m ³

4	Thermal Expansion Co-efficient	7.2	$\mu\text{M}/^\circ\text{C}$
5	Thermal Conductivity	11	watt/m k
6	Specific Heat	950	J/kg K

4.1 RESULTS OF TEMPERATURE DISTRIBUTION

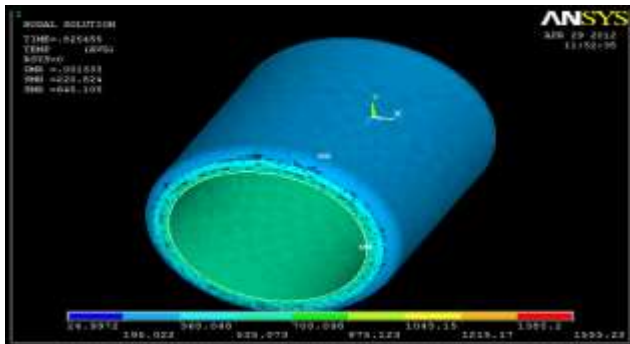


Figure 4.1 Temp distributions in furnace wall after 1 hour for alumina ramming mass

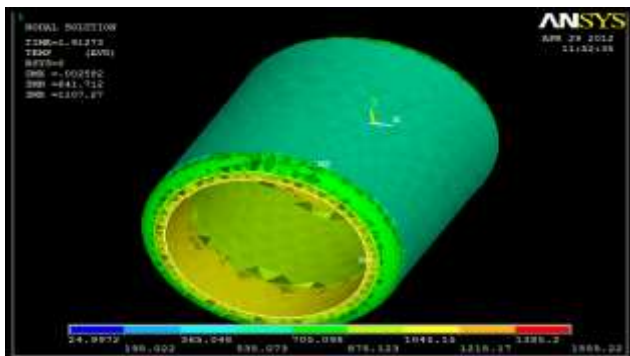


Figure 4.2 Temp distributions in furnace wall after 2 hour for alumina ramming mass

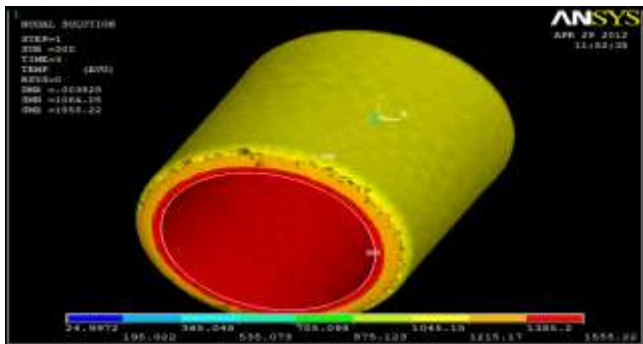


Figure 4.3 Temp distributions in furnace wall after 3 hour for alumina ramming mass

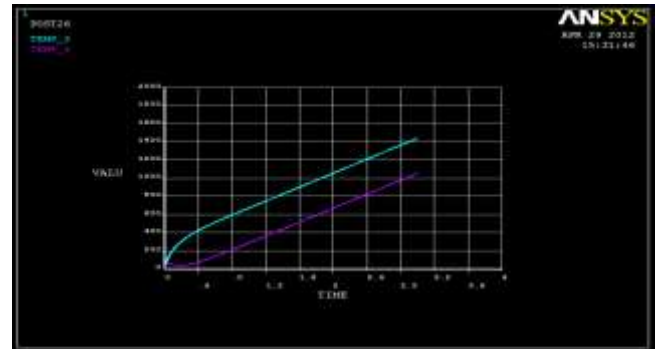


Figure 4.4 Temperature inside and outside furnace wall for alumina ramming mass

4.2 RESULTS OF THERMAL STRESS DISTRIBUTION

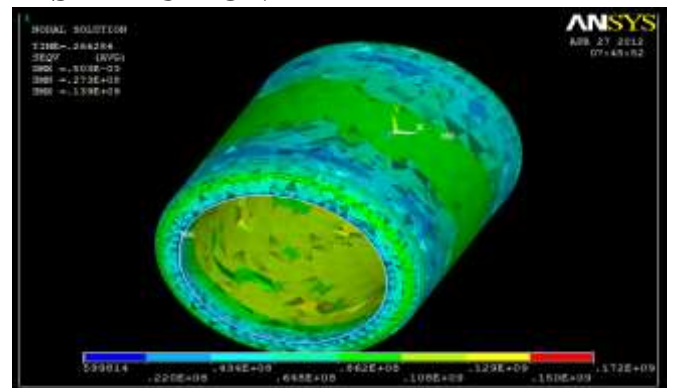


Figure 4.5 Stress distributions in furnace wall after 20 minutes for alumina ramming mass

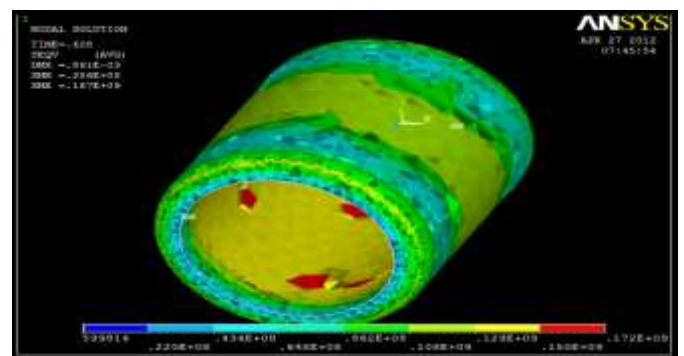


Figure 4.6 Stress distributions in furnace wall after 40 minutes for alumina ramming mass

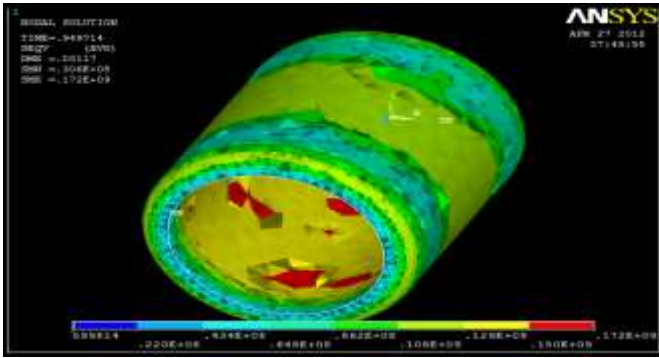


Figure 4.7 Stress distributions in furnace wall after 60 minutes for alumina ramming mass

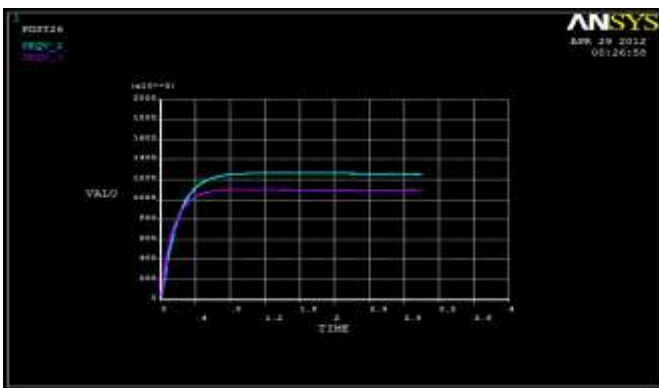


Figure 4.8 Stress outside and inside furnace wall for alumina ramming mass

5.0 PREDICTION OF THERMAL FATIGUE LIFE

5.1 FATIGUE LIFE METHODS [6]

The three major fatigue life methods used in design and analysis are the stress-life method, the strain-life method, and the linear-elastic fracture mechanics method. These methods attempt to predict the life in number of cycles to failure, N , for a specific level of loading. Life of $1 \leq N \leq 10^3$ cycles is generally classified as low-cycle fatigue, whereas high-cycle fatigue is considered to be $N > 10^3$ cycles. The stress-life method, based on stress levels only, is the least accurate approach, especially for low-cycle applications. However, it is the most traditional method, since it is the easiest to implement for a wide range of design applications, has ample supporting data, and represents high cycle applications adequately.

The strain-life method involves more detailed analysis of the plastic deformation at localized regions where the stresses and strains are considered for life estimates. This method is especially good for low-cycle fatigue applications. In applying this method, several idealizations must be compounded, and so some uncertainties will exist in the results. For this reason, it will be discussed only because of its value in adding to the understanding of the nature of fatigue.

The fracture mechanics method assumes a crack is already present and detected. It is then employed to predict crack growth with respect to stress intensity. It is most practical when applied to large structures in conjunction with computer codes and a periodic inspection program.

5.2 THE STRESS-LIFE METHOD

To determine life of any component by Stress-Life Method, we need to find out ultimate strength and endurance limit of the component for the required material.

We know the values of ultimate stress for these all materials. Alumina ramming mass is having ultimate strength of 272 MPa. We can find out Se' from the equation given below or we can say dividing value of ultimate strength.

We know value of ultimate stress of the material.

Material	S_{ut} (MPa)
alumina ramming mass	271

We know the relation between S_{ut} and Se' so that we can find out Se' .

$$Se' = 0.5 * S_{ut}$$

Material	Se' (MPa)
alumina ramming mass	135.5

Endurance Limit Modifying Factors:

We have seen that the rotating-beam specimen used in the laboratory to determine endurance limits is prepared very carefully and tested under closely controlled conditions.

It is unrealistic to expect the endurance limit of a mechanical or structural member to match the values obtained in the laboratory.

Some differences include

- Material: composition, basis of failure, variability
- Manufacturing: method, heat treatment, fretting corrosion, surface condition, stress concentration
- Environment: corrosion, temperature, stress state, relaxation times
- Design: size, shape, life, stress state, stress concentration, speed, fretting, galling.

Marin identified factors that quantified the effects of surface condition, size, loading, temperature, and miscellaneous items. The question of whether to adjust the endurance limit by subtractive corrections or multiplicative corrections was resolved by an extensive statistical analysis of a 4340 (electric furnace, aircraft quality) steel, in which a correlation coefficient of 0.85 was found for the multiplicative form and 0.40 for the additive form.

A Marin equation is therefore written as

$$Se = k_a \times k_b \times k_c \times k_d \times k_e \times k_f \times Se'$$

Where,

k_a = surface condition modification factor

k_b = size modification factor

k_c = load modification factor

k_d = temperature modification factor

k_e = reliability factor

k_f = miscellaneous-effects modification factor

Se'= specimen endurance limit

Se = endurance limit at the critical location of a machine part in the geometry and condition of use

We can find out different as per the guideline given in the Machine Engineering Design by Shigley.[19]

Surface Finish Factor ka	0.86
Size Factor kb	1
Loading Factor kc	1
Temperature Factor kd	0.25
Reliability Factor ke	0.9
Miscellaneous Effects Factor kf	1

Now, we can find out endurance limit for all different materials.

$$Se = ka \times kb \times kc \times kd \times ke \times kf \times Se'$$

Material	Se (MPa)
alumina ramming mass	26.21

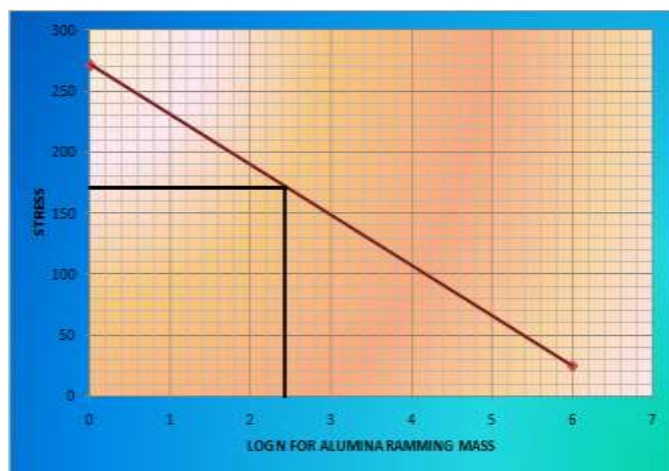


Figure 5.1 Modified S – Log N Curve for Alumina Ramming Mass

6.0 CONCLUSION

Induction Melting Furnaces are highly used now- a-days for melting of different kinds of materials. The problem comes from the Refractory material of losing its thermal properties

within 200-400 hours of lifetime. It will disturb production schedule as it requires time to replace the Induction melting furnace wall. Coupled Field Analysis is done for Induction Melting Furnace Refractory Wall and validation is done with respect to Experimental Results. Coupled Field Analysis is done with respect to Alumina Ramming Mass. Then S – log N Curves are plotted for Life Span Prediction.

We had found life span of different materials like alumina ramming mass 200 cycles. From the results of experimental study and finite element simulation of thermal fatigue failure of induction melting furnace wall, it can be seen that finite element simulation exactly predicts the failure of the induction furnace refractory wall and the definite solution conditions in the finite element numerical calculation are accurate.

The fatigue life of the induction melting furnace refractory wall under thermal fatigue working conditions was predicted using Stress-Life Method by plotting S – log N curves for different materials on the basis of finite element calculations and maximum stress in the induction melting furnace refractory wall. We can use it as a linear to increase lifespan or we can use premixed ramming mass for economical and better working lifespan of induction melting furnace wall.

The accuracy of the fatigue life prediction for the induction melting furnace refractory wall depends upon stress spectrum calculated at the critical point by finite element method and on S-log N fatigue curves prepared from the material properties of different materials.

7.0 REFERENCES

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